### The Field Modelling of Fire in an Air-Supported Structure by

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## ABSTRACT

The field modelling technique for predicting the temperature distribution and smoke movement in enclosures containing a fire source is validated against experiments carried out in a fully instrumented sports building covered by an air supported dome. The building is oval in plan and the dome has an ellipsoidal shape.

A 2MW methanol pool fire located centrally on the floor of the building was used to obtain detailed measurements of temperature at a number of locations. The mathematical model simulates the transient problem in three dimensions using two different finite volume grids. The first grid is a polar cylindrical one with cells partially blocked to simulate features not coincident with grid lines. The second uses a non-orthogonal grid which follows closely the contours of the building. Results are obtained for pre-fire. fire and post-fire conditions and the two grid solutions are compared with experiments.

Qualitative agreement is good throughout and trends are correctly simulated. Quantitative agreement is also good in all areas except in common with earlier studies in the immediate vicinity of the fire source. The body fitted grid solution predicts correctly the lack of stratification due to strong convection along the ceiling.

Keywords: Mathematical model, smoke, body-fitted coordinates, fire

### 1. INTRODUCTION

The mathematical field modelling of fire has now reached the stage where two distinct branches of further development are taking place. On the one hand smoke movement models, already reasonably well validated at least in all except the very largest of buildings [1], are being applied to practical problems for design purposes. This is leading to refinements of the models in the treatment of the physical phenomena involved (eg. the prediction of carbon monoxide [2]) and in the complexity of the geometries being tackled [3]. There are also developments of user-friendly tools for easier use by the non-expert and for incorporation into 'systematic' fire models which fuse developments in the modelling of human escape, life threat and structural response with those of the gas phase.

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On the other hand, and building on these capabilities, is the development of offre spread' models where the incorporation of more realistic radiation models tcan be coupled with the development of semi-empirical treatments of material m 35 ... 18 flame spread [5]. Int me and This paper deals with the further development and validation singlify smoke The vehicle for the study is an experimental air supported movement models. dome in which Nakamura and his colleagues at the Shimizu-Construction Company [6] conducted a number of fire tests. The experimental facility has major dimensions 34m x 28m x 11.6m in-height, being a-one sixth<sup>12</sup> scale model of a proposed sports stadium (200m x 200m x 70m). Even in its, one sixth scale this building represents, as far as the authors are aware, the largest single volume used for field model validation. In addition to the value of the study as further model validation at large scale, it also offers because of the shape of the roof, an opportunity for testing and developing a body-fitted-coordinate version of the field model JASMINE.

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The details of JASMINE have been described in detail elsewhere [1,7,8]. It is based on the general fluid dynamics package PHOENICS [9] as the main equation solver.

The work described here represents the development of a second version of the model to be known as JASMINE II which is based on the latest release of PHOENICS with Improved numerical features and possesses a generalised non-orthogonal coordinate scheme.

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### 2. PHYSICAL MODEL SIMULATED

The experimental building under consideration, constructed by Shimizu Construction Ltd. of Tokyo. Japan is an oval sports stadium with an air supported roof. It was designed as a 1/6th scale model of a proposed full size stadium which would hold a capacity crowd of 60,000 people. The dimensions of the model are, 11.6m in height at the dome apex and 34m by 28m In plan. It was fully instrumented, in particular for the measurement of Figure 1 shows a plan and side elevation together with the temperatures. positions of the thermocouple rakes. positions of the thermocouple rakes. All of the thermocouple rakes. The dome was held aloft by an overpressure of 25mm water guage. The dome was held aloft by an overpressure of 25mm water guage. E ti ni pressure was supplied by compressors located in two internal "sheds", shown inti Air at a controlled temperature was supplied through a series of the figure. slots in a ring located around the periphery of the stadium, at a height of 7mar Air supply was computer controlled and divided into four quadrants. 21 A continuous supply was necessary to compensate for air, losses through, the dome and walls. The dome material was constructed for lightness using an .8 mm glass fibre skin. Air gaps exist along the various sections of the roof Nineteen fire tests were conducted, with a fire source placed at various locations on the floor. In the present calculation one of these with a centrally located fire source is modelled. The fire source comprised 4 square trays of methanol.<sup>9</sup> arranged in a square, with a 2m dimension. The fire duration was 14 minutes with an estimated peak heat release rate of 2.2MW and to inclusion

during and after the fire had been Measurements were taken before, extinguished, and these were used in the comparison presented here. and all One further point to mention, is that, throughout the experiment two opposing quadrants only supplied the air for the dome. Table at shows the external air supply temperature and mass flow rates and also the amount of air recirculated for temperature control. ne me gole his 116 311

is I. comined a row you' Table 1: Air Conditioning Settings or a

Quadrant	2	4
Total mass flow	، ج ب ج	nus v in
(kg/s)	3.023 ,	2,312
Supply Temp	ally 3	BELL I
(K)	31.9	13.3
Recirculation rate	<u>_</u> 693	, a
(kg/s)	2,124	sh 1.625
Velocity (m/s)	<mark>ی 0.696</mark>	<sub>3</sub> 0.5

" avah As the two quadrants blow air 81 B + 15 at different temperatures, they induce recirculation in the flow. This recirculation is important as it dictates the way smoke spreads during the first few minutes of the fire, an important consideration for smoke detection device placement. For this reason, this pre-fire condition was 15 calculated first. RISC

#### THE MATHEMATICAL MODEL З.

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ŕ: The mathematical method used 5 in this investigation is essentially the same as the one used in a previous investigation by the present research team and will not be discussed in detail here. The interested reader should consult References [7,8,10] for full It is sufficient to say here details.

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that equations for the conservation of 1510 1enthalpy and <sup>G</sup> fuel<sup>21X</sup> mixture fraction. mass, en momentum, to desribe the concentration of smoke, are solved throughout the calculation domain either on a cylindrical polar or a body-fitted coordinate grid. The two equation k-c model is used to describe turbulent mixing. One important addition is the treatment of the effects of buoyancy on the source term Se in the e equation in The modification recommended by Bos et al [11] accounts for the varying, direction of the gravitational vector on the ceiling jet following the contours of

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Fig 1: Experimental Dome Showing

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### c) View from east

a curved roof.

boundary

$$S_{c} = \frac{\epsilon}{k} \left[ (G_{k} + G_{B}) \tanh^{2} \frac{e^{-w}}{(u^{2}+v^{2})^{1/2}} \right] C_{1}^{i691} C_{2}^{i} \rho_{e}^{i} \rho_{e}^$$

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where Gk and GB are the generation terms for turbulent kinetic energy due to shear and buoyancy respectively which take their usual form [10]. A four flux model was used to describe radiant heat exchange between adjacent cells. The four fluxes used were aligned along the longitudinal (vertical) and radial ordinates. 31 - an ya The treatment of othe body-fitted-coordinate ligrid has been described by 2 d + 0 d d d 2 d - 1 d - 1 d - 1 d - 1 d - 1 d - 2 d Markatos et al [12]. 1.98 . . .

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3.1	Boundary Conditions	00	be Ht	13	f		:• .
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These areaentered in the form of additional source"terms to represent friction and heat transfer at the boundaries due to convection, conduction and colis radiation. For friction, log-law wall functions are sused to act as sinks of momentum in

the appropriate velocity direction. Similarly for the convective heat transfer at the walls and ceiling the so called Reynolds vanalogy between momentum and heat transfer is employed. This requires the wall temperatures which are calculated from a balance at the

- 
$$U_{em}$$
  $(T_m - T_e)$  +  $U_{mp}$   $(T_p - T_m) = \rho \frac{Cp_w B (T_m - T_m^*)}{\Delta t}$  (1)  
where the subscripts e. m and p refer to external, wall mean and international int

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$U_{mp} = \frac{1}{1 + B}$	and $U_{em} = \frac{1}{1 + B}$	(2)
hi 2k <sub>C</sub>	aho ta 2k	с

B is the wall thickness, kc the conductivity and h the convective heat transfer coefficient. 11)

Tm\* is the "old" wall mean temperature calculated at the previous time step. A minimum convective wall heat transfer limit is set at the natural convection limit given by . All

$$Nu = 0.13 Ra^{1/3}$$

(3)

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where the temperature gradient would lead to natural convection at the boundary. Other and 30 the track it as 11 The boundary condition for the radiation fluxes takes into account the wall emissivity and temperature.

The fire is entered as a heat source per unit area and also as a source of mass. This mass also enters the conservation equation representing the concentration of burned fuel in the building.

Finally, the mass and momentum entering the two active air conditioning quadrants is introduced, and air leakage resistances adjusted to achieve the required over pressure in the dome.

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### child by 3.2 The Solution Method

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by the JASMINE/PHOENICS code, with JASMINE being the problem specifying element and PHOENICS comprising the equation solver. Details relating to the solution and formulation of the finite-domain equations and the algorithms used are given in [7].

The finite domain representation of the problem is shown in Figure 2 in cylindrical polar and body-fitted coordinate form. A total of 2860 computational cells is used in the polar grid case compared to 1200 in the body fitted coordinate case. However, in the polar grid case about 15% of the cells are unused, as they lie; outside the calculation domain and are "blocked off". The number of cells used, was chosen as a compromise between accuracy and economy and therefore no claim of grid independence is made. The comparison with experiments was deemed to be satisfactory nevertheless and due to uncertainties in experimental measurements it was felt that further refinement at this stage in terms of validation would be unnecessary.





### 4. RESULTS OF CALCULATION

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4.1 Calculations Performed

The calculation was split into three parts:

(i) a pre-fire steady state calculation;

(ii) a transient calculation for the duration of the fire; and 30° durate a post-fire transient calculation following the burn-out of the fuel; 100 durate the burn of the fuel; 100 durate 100 durate dur

Each one of the transient cases involved a number of time-steps and the result of each timestep was stored in the form of a field of velocities. pressure, temperature and fuel concentration and then plotted using a The amount of results so produced is very large and in the post processor. space available only a small sample can be presented, in condensed form. In the main fire transient (ii), the heat source was modelled by a linear growth over the first 90 seconds to the maximum constant level of rate of heat release of 2.2MW over the next 5.5 minutes and a linear decay from this level to zero for a further four minutes as the fuel becomes exhausted. The numerical timesteps were chosen to ensure timestep independence in the initial critical stage and to resolve the spreading of the combustion products under These varied between 2 seconds and 10 seconds in the initial the ceiling. increasing ramp, to 30 seconds for the steady and decaying phases of the fire.

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Results are presented for velocity vectors and contours of temperature and fuel mixture fraction in Figures 3. 4 and 5. In Figures 6 to 8 for direct comparison with the experimental data. temperatures on thermocouple rake 3 are displayed.

The pre-fire condition is shown in Figure 3. It shows horizontal sections below the roof, at the plane of the air inlets and at the floor, and also a section through the plane crossing the top of the control sheds. The dominant feature of the flow is a large scale recirculation caused by the air-conditioning inlets which supply air at different temperatures. hence, air rises in the vicinity of quadrant 2 while it falls near quadrant 4. Maximum velocities lie close to the roof and reach 0.5 m/s. Due to this circulation the temperature overall is quite uniform at a mean value of  $17^{\circ}C$ .



Fig 3: Prefire conditions at Levels of Air Inlet. Top of Control Shed and Floor

### 4.2 Fire and Post-Fire Translent Results

The development of the transient calculation for the fire's duration is depicted in two different ways. Figure 4 shows a series of sections through the centre of the building viewed from the east. These sectional views show velocities and temperature contours at selected time intervals throughout the fire and just into the post-fire condition. The spread of combustion products just under the celling is shown during the initial growth phase of the fire in Figure 5 together with velocity vectors.

The initial prefire circulation in the stadium leads to an asymmetry during the early stages of the fire. By about two minutes this is largely overcome and the temperature contours outside of the rising plume become horizontal. As the fire diminishes in rate of heat release over the period from seven to eleven minutes the plume thickens as the buoyancy reduces. At ten minutes the plume is very wide and there are large scale circulations on either side of it. By 11.5 minutes the fire is out, the buoyant plume has disappeared and an inverse plume appears at the roof due to ambient cooling. The temperature throughout the volume is fairly uniform.

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### 4.3 Smoke Spread

Figure 5 shows the spread of combustion products and the velocities below the roof fabric. The flow field initially is that of the pre-fire condition. After 14 seconds the plume first touches the ceiling and as time proceeds, the flow field changes from a lateral flow to a radial impact jet. The initial condition leads to asymmetric contours which persist over most of the first two The values shown on the contours are for the fuel mixture fraction, minutes. which can be related to product concentration by the use of the reaction At 14 seconds, the 0.0005 contour reaches the roof while at 70 equation. seconds it reaches the stadium wall, a mean distance of 17m. A rate of spread of the products of combustion of about 0.3m/s is therefore indicated. At two minutes from ignition (not shown) smoke concentrations are up on this at the centre of the roof by a factor of 5.

### 5. COMPARISON WITH EXPERIMENT

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### 5.1 Scope of Experimental Data

Five vertical arrays of thermocouples where placed in various positions in the stadium as depicted in Figure 1. Readings were then logged on a micro-computer every 30 seconds. In addition to temperatures, a velocity probe was placed close to the stadium wall which recorded a maximum vertical velocity of the order of 0.2 m/s.

Smoke was released into the normally clean-burning fire plume and observed on a video camera to provide a qualitative indication of smoke spread.

The thermocouple readings are the most substantial and reliable form of data from the transient test and were therefore used here to compare with predictions.

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Fig 5:

Velocity and Fuel Mixture Fraction. Contours Under the Celling in the Growth Phase of the Fire

# 5.2 <u>Temperature History</u>

Figure 6 shows predicted and experimental temperature behaviour during the fire transient for thermocouple array positions 1 and 3 for the body-fitted grid. The rate at which energy is released is clearly demonstrated in position 1, directly above the fire. The dotted lines indicate the envelope of experimental values on the rake, and the solid lines the prediction at various heights.

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The uppermost trace lies closest to the fire and demonstrates clearly the ramp-up, constant burning rate and ramp-down sections. Although the predictions match well the experiments in shape, the predicted temperatures as would be expected from earlier studies (eg. Reference [1]) are low compared to the experiment due to the coarseness of the grid at the source and simplified representation of heat release. However, away from the source the agreement is reasonably good.

Figure 7 shows the temperature rise with height measurements and predictions for both grids used, at rake 3, 11m from the fire. Both grids show reasonable agreement with the data except near the ceiling where the bfc grid seems far superior. The cylindrical polar grid shows an apparently well mixed celling layer not supported by the experiment nor the bfc calculation. Clearly the bfc grid better represents the shape of the ceiling as so perhaps this result is not surprising.

Finally as a summary of the practical implication of the experimental and, numerical study a plot of the height of the 10 degree rise contour with time is shown in Figure 8 for rakes 3, 4 and 5. The agreement at head height (say 2m) is reasonably good except) for the wall station where the model slightly underpredicts the time for this temperature contour to arrive.

### 6. CONVERGENCE AND COMPUTING REQUIREMENTS

### 6.1 Convergence

In 14-1

The coupling of momentum and energy fields through density, combined with the effects of buoyancy leads to non linear behaviour typical to high Rayleigh number situations. Hence, inertial under-relaxation was necessary for convergence coupled with a large number of iterative sweeps per timestep. Similar relaxation was used in the polar and bfc grids. Typically 100 sweeps were sufficient for convergence with cumulative volumetric errors being 1.5 m<sup>3</sup>/s compared to a domain volume of 8000 m<sup>3</sup>.

ele on all astu 6. 2ne <u>Computing Requirements</u>

b Computing requirements are summarised below in Table 2 for cylindrical-polar and body-fitted grid simulations. Computations were performed on a Concurrent 3280 super mini computer.

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Flg 6: Predicted and Measured Flg. 70 STemperature Rise Predicted Temperature Histories at Stations by Both Grids and Measured at 1 and 3 with the bfc Grid Station 3 J XE-0 5~ NUL SC JIN 109-112 9". e.

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#### 7. CONCLUSIONS

Reasonable validation of these computational fluid dynamic field models has already been demonstrated for the treatment of smoke movement problems.

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The study conducted here adds a large volume building to those already satisfactorily validated and in addition has provided a very important opportunity to develop the body fitted coordinate capability to the model JASMINE. This is

an important development for the treatment of many building designs being proposed in modern architectures.

In common with earlier studies, comparisons with the data provided by the Shimizu Corporation show good agreement remote from the fire source but poor agreement in its immediate vicinity. This is to be expected with the coarse grid prescription of the source.

The calculations demonstrate that in a building of this type the familiar well mixed hot layer bath does not form, mainly due to a strong radial jet emanating from the point of plume impact on the celling.

The body fitted grid represents the resulting celling boundary layers correctly and hence yields the best result without sacrifice in economy.

The number of time steps and cells used in this work was kept at economical levels to allow many runs during model development. It is not therefore expected that the results are either grid independent or timestep independent. Good agreement for all practical purposes has nevertheless been achieved, adding confidence in the usefulness of the method in fire predictions to this scale of enclosure.





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Research Establishment, Department of the Environment. UK. It is contributed by permission of the Director of the Building Research Establishment. rch. sons lith 7.1 REFERENCES Most Stomer د 91 S 9. winds num 1. Cox G. Kumar S and Markatos N C 91.. 1 11 14 1. Some Field Model Validation Studies ottor ( 21) Bi First<sub>gn</sub> International eub Symposium on Fire Safety Sclence. p159. 35 Hemisphere. 1986 2001 Moss J B. Cox G and Kumar S 91 to to 1 4 551 30 'Laminar Flamelet Models of Fire - the Prospect' bet Isolon Fundamental Aspects of Polymar Flamability. p53. Institute of Physics. 4 ອາດ:ອາເ. **1987**. 1.10 M -) Galea E R and Markatos N C ... or of Aircraft Cabin Fires 18d 'A Review of Mathematical Mgdelling of Aircraft Cabin Fires 18d 3. ). 91: . Y me · + Appl. Math. Modelling...11, 162. 1987. 'onild' 4. Shah N G 'New Method of Computation of Radiant Heat Transfer in Combustion Chambers' odin 1 5 .10 PhD Thesis. Imperial College. London, 1979. 5. Quintiere J G. 'A Simplified Theory for Generalizing Results from a Radiant Panel Rate of Flame Spread Apparatus. Fire and Materials' 5. 52. 1981 6. H Nakamura 'Fire Behaviour Tests in an Air-Supported Structure' 1 4 private communication, 1986. 7. Markatos N C and Cox+G+++ Hydrodynamics and Heat Transfer in Enclosures Containing a Fire Source' Physicochemical Hydrodynamics, 5, 53, 1984. 8. Markatos N C and Pericleous K A 'An Investigation of Three-Dimensional Fires in Enclosures' Fire Dynamics and Heat Transfer, p115, ASME, 1983 9. Spalding D'B 'A General Purpose Computer Program for Multi-Dimensional One- and Two-Phase Flow' Mathematics and Computers in Simulation, 23, 267, 1981 10. Cox G and Kumar S 'Field Modelling of Fire in Forced Ventilated Enclosures. Combustion Science and Technology Occree rilse 10 - 7 52. 7. 1987. Bos W G. Vanden Elsen T. Hoogendoorn C J and Test F L 11. 'Numerical Study of Stratification of a Smoke Layer in a Corridor' Comb Science and Technology 38, 227, 1984. 12. Markatos N.C. Malin M R and Tatchell D G. Computer Analysis of Three-Dimensional Turbulent Flows Around Ships Hulls' pnituffec lest Mech Eng 194, 239 and 1980. - · · · · · 16. 1

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